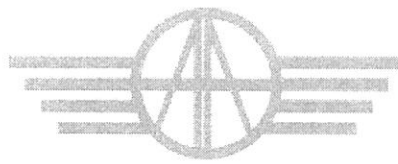




REPUBLIC OF ALBANIA



AUTORITETI I AVIACIONIT CIVIL

ALBANIAN CIVIL AVIATION AUTHORITY

AIRWORTHINESS DIRECTIVE

ACAA-DFS-AD-No.010

Issue: 01, Revision 00

Date: 12.07.2024

Approved by:

Maksim Et'hemaj

Executive Director of Albanian Civil Aviation Authority



0.1 Record of Amendments

The table below describes the dates and reason for the different amendments of the current procedure. A vertical black line on the left-hand side of the page identify the changes with the previous version.

Issue No.	Revision No.	Date	Amended by	Reason
01	00	07.11.2023	SAW	Initial Issue

0.2 Revision table

Page #.	Issue No.	Revision No.	Date	Edited by

1. Name of the AD:

AD No.: 2024-0066, ATA 05 – Time Limits / Maintenance Checks – Safe Life Airworthiness Limitations Items – ALS Part 1 – Amendment.

2. Full Description of the AD:

The airworthiness limitations for the Airbus A320 family aeroplanes, which are approved by EASA, are currently defined and published in the Airbus A318/A319/A320/A321 ALS document. These instructions have been identified as mandatory for continued airworthiness. The Safe Life Airworthiness Limitations Items (SL-ALI) are specified in ALS Part 1.

Failure to accomplish these instructions could result in an unsafe condition.

Previously, EASA issued AD 2022-0082 to require accomplishment of all SL-ALI maintenance tasks as described in ALS Part 1 at Revision 08.

For the reason described above, this AD retains the requirements of EASA AD 2022-0082, which is superseded, and requires accomplishment of the actions specified in the ALS.

3. Issued and Effective Dates:

Issued: 08 March 2024

Effective: 22 March 2024

4. Full List of Aircraft Affected:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers.

5. Description of How It Can Be Resolved:

Maintenance Tasks and Replacement of Life Limited Parts:

(1) From the effective date of this AD, replace each component before exceeding the applicable life limit, as specified in the ALS (see Note 1 of this AD), as applicable to aeroplane model and depending on aeroplane configuration.

Note 1: For the purpose of this AD, the thresholds and intervals, as defined in the 'Compliance Time' pages of the ALS, include specific compliance times for certain tasks.

AMP Revision:

(2) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the limitations described in the ALS, as applicable to aeroplane model and depending on aeroplane configuration.

Credit:

(3) If, before the effective date of this AD, the AMP has been revised to incorporate the life limitations as specified in a previous revision of the ALS, that action ensures the continued accomplishment of those limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations as specified in the ALS, as applicable to aeroplane model and depending on aeroplane configuration, within the compliance times (see Note 1 of this AD) as specified in the ALS to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and more restrictive tasks and limitations as specified in the ALS, as applicable to aeroplane model and depending on aeroplane configuration, into the AMP to comply with paragraph (2) of this AD.

Recording AD Compliance:

(4) When the AMP of an aeroplane has been revised as required by paragraph (2) or (3) of this AD, as applicable, that action ensures continued accomplishment of the tasks as required by paragraph (1) of

this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (2) or (3) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.

Ref. Publications:

Airbus A318/A319/A320/A321 ALS Part 1 Revision 09 dated 06 November 2023.

The use of later approved variations or revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 01 February 2024 as PAD 24-014 for consultation until 29 February 2024. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – 1IASA; E-mail: account.airworth-eas@airbus.com.

For full compliance please refer to :

https://ad.easa.europa.eu/blob/EASA_AD_2024_0066.pdf/AD_2024-0066_1